# Condensation Trails in Trajectory Optimization

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*Abstract*—Contrails are one of the driving contributors on global warming, induced by aviation. The impact of contrails on global warming is subject to large uncertainties of more than 100 %. In detail, condensation trails might even change the algebraic sign between a cooling and a warming effect in an order of magnitude, which is comparable to the impact of aviation emitted Carbon dioxides and Nitrogen oxides. This implies the necessity to granularly consider the environmental impact of condensation trails in single trajectory optimization tools. The intent of this study is the elaboration of all significant factors deciding on the net effect of single condensation trails. Possible simplifications will be proposed for a consideration in single trajectory optimization tools. Finally, the effects of the most important impact factors, such as latitude, time of the year and time of the day, wind shear, atmospheric turbulence and their consideration in a multi-criteria trajectory optimization tool are exemplified. The results can be used for an arbitrary trajectory optimization tool with environmental optimization intents.

*Keywords: Contrails, Trajectory Optimization, Energy efficiency, Weather impact*

#### I. INTRODUCTION

Condensation trails (contrails) summarize a type of human induced clouds, developed behind aircraft due to condensation of exhausted water vapor emissions and ambient humidity around exhausted soot particles and atmospheric condensation nuclei [1] in a cold ambient atmosphere, satisfying the Schmidt-Appleman-criterion [2, 3]. In an ice supersaturated environment, those artificial ice clouds form into long living cirrus clouds called "Cirrus homogenitus" as defined by the World Meteorological Organization [4].

In the Earth-atmosphere energy budget, contrails act like a barrier [5–7]. They scatter incoming shortwave solar radiation back to the sky (resulting in a cooling effect) and they absorb and emit the outgoing longwave terrestrial radiation back to the Earth's surface (yielding a warming effect in the lower layer of the atmosphere) [6, 8–10]. The dominating effect may be defined as radiative forcing RF, as the imbalance in the radiation budget of the Earth-atmosphere system (considering the instantaneous response of the stratosphere). The contribution of contrails to global warming is a subject to uncertainties and depends on flight performance, weather conditions and time.

Latest combinations of several approaches to model a global impact of contrails on global warming summarize a warming net effect of  $RF_{\text{Contrail}} = 0.05 \text{ W m}^{-2}$  with uncertainties between  $-0.02$  and  $+0.15$  W m<sup>-2</sup> for 2010 [6]. Single studies result in a larger environmental impact of contrails including negative (cooling) net effects (e.g.  $RF_{\text{Contrail}} = -0.007 \text{ to } +$  $0.02 \text{ W m}^{-2}$  for 2005 [9]). Compared to the summed effect of one year's aviation emitted Carbon dioxide  $CO<sub>2</sub>$  and Nitrogen oxides NO<sub>x</sub> which is  $RF_{\text{CO}_2,\text{NO}_x} = 0.04 \text{ W m}^{-2}$  [6], the impact of contrails on global warming is no longer negligible [6, 11], because it might exceed those effects of  $CO<sub>2</sub>$  and  $NO_x$ , although contrails are only formed during 10% of the flight, on a global average [12]. Thus, the need for contrails to be considered in trajectory optimization becomes indisputable.

The impact of single condensation trails on trajectory optimization has been analyzed by Gounou et al. [13] and Forster et al. [14] focusing on the importance of large solar zenith angles during sunset and sunrise. In an application of a Monte Carlo code for photon transport, Forster et al. [14] already considered effects like multiple scattering, but in a coarse spatial grid and ignoring the impact of flight performance on the optical properties of the contrail. Detailed studies on all significant impact factors have been elaborated by Rosenow [11]. In the current study, the results have been simplified and harnessed for trajectory optimization. All other research interests known to the authors concentrated on the effect of contrails on a global scale. Using global climate models and historic air traffic data, reliable estimations of the global contrail radiative forcing for the year 2000 of  $RF_{\text{Contrail}} =$ 0.03 ( $-0.01$  to  $+0.08$ ) W m<sup>-2</sup> [9] have been improved for 2010 to  $RF_{\text{Contrail}} = 0.02 (-0.01 \text{ to } +0.03) \text{ W m}^{-2}$  [6] considering an increased traffic distance by 22 % between 2005 and 2010. For 2002, Burkhardt and Kärcher [10] estimated  $RF_{\text{Contrail}} = 0.03 \text{ W m}^{-2}$  of contrails and contrail cirrus within a global climate model.

Using satellite data, the Adjusted Forcing  $AF$  as imbalance of the Earth-atmosphere energy system after stratospheric temperatures has been adjusted to regain a radiative equilibrium in the stratosphere (assuming zero further radiative heating rates) can be calculated [15, 16], considering a completed transition of contrails into cirrus. Herewith, the diurnal cycle of contrails and cirrus and differences in regions with low and high air traffic demand can be distinguished [16]. Using satellite data of 2006,  $AF_{\text{Contrail}} = 0.045$  to 0.075 W m<sup>-2</sup> has been quantified for contrails and contrail-induced cirrus. A combination of modeled and satellite data-based estimates and a consideration of uncertainties in spreading rate, contrails optical depth, ice particle shape and radiative transfer [17] and accounting for the ongoing increase in air traffic, a contrail and contrail-induced cirrus  $AF<sub>Contrail</sub>$  for the year 2010 of  $AF_{\text{Contrail}} = 0.05$  (0.02 to 0.15) W m<sup>-2</sup> is widely accepted [6].

A consideration of contrails in trajectory optimization tools if any, has been found as constant value for trajectories through ice-supersaturated regions [18–22]. This consequent avoidance of contrail formation does not, however, lead to a holistically optimized trajectory [23]. Specifically, considering single contrails with cooling effects on the Earth-atmosphere system are completely misunderstood in those approaches.

# II. INDIVIDUAL CONTRAILS IN TRAJECTORY **OPTIMIZATION**

The assessment of single contrails in trajectory optimization may be performed in six steps. The procedure is shown in Fig. 1.

## *A. Atmosphere GFS Weather Data*

First, detailed, weather information of the environment around the aspired route is required. Modeled and gridded global forecast data from the Global Forecast System GFS, provided every six hours in Grib2 format by the National Oceanic and Atmospheric Administration NOAA with a spacial resolution of 0.25 degrees at 18 pressure levels is a good compromise between computational effort and accuracy. For trajectory optimization and the estimation of the conditions of contrail formation, vertical profiles of temperature  $T$  [K], pressure p [Pa], density  $\rho$  [kg m<sup>-3</sup>], horizontal wind components u  $[\text{m s}^{-1}]$  and v  $[\text{m s}^{-1}]$ , vertical velocity w  $[\text{Pa s}^{-1}]$ and the relative humidity  $rH$  [-] along the whole route are taken into account.

# *B. Flight Performance*

Together with typical input variables for trajectory optimization, such as city pair, aircraft type, engine type, payload, optimization function (i.e. minimum fuel burn, minimum time of flight, minimum contrail impact or multi-criteria optimization), a trajectory optimization model with implemented key performance assessment can be used for the calculation of the optimum vertical and lateral path. Here, we use the validated simulation environment TOMATO [24–26] which includes the aircraft performance model COALA [27, 28] for vertical optimization and for the quantification of the emissions. In TOMATO, the trajectory is optimized iteratively by assessing each interim solution regarding several key performance indicators (KPI), contrails amongst others [29]. For comparability,



Figure 1. Data flow diagram of all sub models to calculate the radiatitive forcing of individual contrails.

each KPI is transformed into monetary values. For the evaluation of the contrail, the conditions on contrail formation, the fulfilled Schmidt-Appleman-criterion [2, 3] depending on true air speed  $v_{\text{TAS}}$ , thrust F, fuel flow  $\dot{m}_f$  [kg s<sup>-1</sup>], specific combustion heat of kerosene  $Q = 43 \text{ MJ kg}^{-1}$ , emission index of water vapor  $EI_{\text{water}} = 1.24 \text{ kg kg}^{-1}$  [11] and the ambient ice supersaturation [30, 31] as function of the relative humidity and the vapor pressure over ice [11] are observed each second.

## *C. Contrail Life Cycle*

In the case both criteria are satisfied, T, p,  $\rho$ , u, v, w, rH, longitude lon [°], latitude lat [°], altitude  $z$  [m],  $\dot{m}_f$  [kg s<sup>-1</sup>] and time  $t$  [s] are provided by COALA for the "Contrail Life Cycle" model [11, 32] of each time step, a contrail is induced. For the initial dimensions, a wake vortex model [33] is applied and calibrated for each aircraft type, implemented in COALA. The initial decay of the contrail in the vortex regime strongly depends on atmospheric turbulence (compare Fig. 2, right), which is approximated by the energy dissipation rate  $\varepsilon$  [m<sup>2</sup> s<sup>-3</sup>] as conversion of kinetic energy due to molecular friction per unit mass and per time into thermal energy [34].  $\varepsilon$ is calculated assuming a lognormal distribution of turbulence in the lower troposphere and upper stratosphere and a linear correlation between a logarithmic diagnostic turbulence value, such as the vertical velocity  $w$ , provided by the GFS [35, 36]. Furthermore, wind shear sh  $[s^{-1}]$  (Fig. 2, left) as difference in wind velocity u and v  ${\rm [m\,s^{-1}]}$  between two altitudes  $\Delta z$  [m] strongly influences the 2D sheared Gaussian plume model for contrail dispersion [11, 32].  $\Delta z$  is called shear layer and depends on the maximum differences in wind velocity between two altitudes. sh is also calculated from the

provided weather data. The sheared diffusivity  $D_s$  [m<sup>2</sup>s<sup>-1</sup>] is assumed to be in the range of the square root of the vertical  $D_v$  [m<sup>2</sup>s<sup>-1</sup>] and horizontal diffusivity  $D_h$  [m<sup>2</sup>s<sup>-1</sup>]:  $D_s \approx \sqrt{D_v D_h}$  [37], but  $D_s \leq \sqrt{D_v D_h}$  [11]. Assuming a soot emission index of  $EI_{\text{soot}} = 0.04 \text{ g kg}^{-1}$  kerosene [38] and a proportional share of ice particles in the contrail [38], all variables, impacting the optical properties of the contrail can be provided to the "Contrail Life Cycle" model. In fact, these are the horizontal  $\hat{\sigma}_h(t)$  [m<sup>2</sup>], vertical  $\hat{\sigma}_v(t)$  [m<sup>2</sup>] and sheared  $\hat{\sigma}_s(t)$  [m<sup>2</sup>] components of the contrail diffusivity variance  $\hat{\sigma}(t)$  [m<sup>2</sup>], the contrail cross section  $CCS(t)$  [m<sup>2</sup>], the ice water content  $IWC(t)$  [kg m<sup>-3</sup>] as total amount of ice mass per volume contrail, the number of ice particles  $n_p$  (hereafter called ice particle number density) [m<sup>−</sup>3] and the projected particle area  $A_p$  [m<sup>2</sup>]. The impact of sh,  $\varepsilon$  and  $v_z$  [m s<sup>-1</sup>] on the particle radius and the contrail life time is shown in Fig. 2 and 3, respectively. The vertical wind speed  $v<sub>z</sub>$  is calculated from the mean divergence of the horizontal wind speed, averaged along the vertical axis between ground and flight level [39]. Thereby, negative values indicate upward wind speeds. On average, values of  $\bar{v}_z = -0.005 \text{ ms}^{-1}$ are calculated. The vertical wind velocity is two orders of magnitudes smaller than the horizontal wind velocity and a vertical upwind velocity of  $v_z = -0.005 \text{ ms}^{-1}$  is realistic in stable stratification [40]. Assuming this vertical upward speed and a sedimentation of the ice particles following Stoke's law, the contrail sediments as soon as the ice particle radius exceeds values of  $r_p \approx 6 \mu m$  [11].



Figure 2. Increasing ice particle radius and decreasing contrail life time with increasing wind shear sh (left) and with decreasing atmospheric turbulence (right).

Note, the available ice water content is distributed equally to the available number of ice particles per contrail volume. A decreasing fuel flow causes less number of soot particles (as primary condensation nuclei), but larger ice particles. However, the optical properties of contrails stronger depend on ice particle size as on ice particle number density [11]. From this follows, that flying slowly through ice-supersaturated regions causes more radiative effective contrails. Furthermore, according to the Schmidt-Appleman-criterion [2, 3], the higher the overall engine efficiency, the lower the exhaust gas temperature, the higher the critical atmospheric temperature  $(T_{\text{LC}})$ for contrail formation [41]. In general, optimized air speeds

for highly efficient conditions of combustion (i.e. minimum fuel flow) refer to low values of  $TAS$ . Both facts support the thesis that cold, ice-supersaturated regions should rather be flown through with high speeds.



Figure 3. Impact of vertical wind speed  $v<sub>z</sub>$  on particle radius and contrail life time. The stronger the upward wind, the longer the contrail remains in the ice-supersaturated region.

#### *D. Contrail Optical Properties*

The fourth step is the "Contrail Optical Properties" model. Depending on the geometrical and microphysical properties of the contrail, the radiative extinction due to scattering, absorption and emission within the contrail is calculated running a Monte Carlo Simulation to consider multiple scattering events which are likely, especially for large solar zenith angles  $\theta$  [rad] [11]. Therefore, Beer's law

$$
\frac{I_{\lambda}(s_1)}{I_{\lambda}(s_2)} = \exp\left[-\int_{s_1}^{s_2} -Q_e A_p n_p(s) \,ds\right]
$$
 (1)

is used, where  $I_{\lambda}(s_2)$  and  $I_{\lambda}(s_1)$  denote the original and the extinguished wavelength specific radiation of solar intensities  $[W \, m^{-2} \text{sr}^{-1}]$  and of terrestrial irradiances  $[W \, m^{-2}]$  (compare Fig. 4).  $Q_e(s)$  denotes the extinction efficiency [-] and depends on wavelength, particle size and shape [11, 42] and  $A_n$ denotes the projected particle area [m<sup>2</sup>].  $Q_e(s)$  and  $A_p(s)$  are not constant within the contrail and depend on the position s. In the "Contrail Optical Properties" model,  $\frac{I_{\lambda}(s_1)}{I_{\lambda}(s_2)}$  (1) is interpreted as number ratio of extinguished photons, regardless of the amount of radiation irradiating the contrail (compare Fig. 4) [11, 43].

The extinction of photons is calculated for each direction in space individually, considering ice a particle shape dependent and wavelength dependent probability of an extinguishing event. The latter is described by the absorption and scattering efficiency  $Q_a(s)$ ,  $Q_s(s)$ .  $Q_s(s) + Q_a(s) = Q_e(s)$  are parameterized by Wyser et al. [42] and Yang et al. [44]) as function of wavelength, ice particle size, shape and density, which in turn are provided by the Gaussian plume model (compare Fig. 4).

This "Contrail Optical Properties" model provides weighted number ratios  $S_i(\lambda, d\omega)$  of extinguished photons per meter contrail, per wavelength and per time step of the contrail life cycle (Fig. 4). For each direction of incoming photons,  $S_i(\lambda, d\omega)$  [m] is calculated by



Figure 4. Application of Beer's law to a "Contrail Radiative Forcing" model.



Figure 5. Geometry of the "Contrail Radiative Forcing" model. The contrail is described by a sheared Gaussian plume. The contrail optical properties further depend on the direction of incoming solar intensities and terrestrial irradiances. The circular solution space is defined by the radius  $w_{\text{in}} = 6\sigma_h$ 

$$
S_i(\lambda, t, d\omega) = \frac{N_{\text{out}}}{N_{\text{in}}} w_{\text{in}} \sin \alpha,
$$
 (2)

where  $N_{\text{out}}$  and  $N_{\text{in}} = 10^7$  denote the number on outgoing and incoming photons, respectively.  $w_{\text{in}} = 6\hat{\sigma}_h(t)$  denotes the irradiated width of the contrail,  $\alpha$  defines the angle between the length axis of the contrail and the incoming photons (compare Fig. 4). This extinction strongly depends on the direction of irradiation (i.e.  $\alpha$ ). The longer the travel distance of photons through the contrail (i.e. the larger  $\alpha$ ), the higher the probability that an extinguishing event takes place. With increasing travel distance and with increasing  $\alpha$ , the number of outscattered photons increases, although a dominating forward scattering is expected [42, 44, 45]. From this follows, that during horizontal photon transport during sunrise and sunset more photons will be scattered in the opposite hemisphere, compared to a vertical photon transport at noon. However, the contrail radiative extinction further depends on the power with which the contrail is irradiated, which will be maximum at noon and minimum at night. This power is calculated separately in the next step.

## *E. Atmospheric Radiative Transfer*

The fifth step of the approach provides wavelength and angular specific solar intensities and terrestrial irradiances, at the position of the contrail. This step is necessary, because the radiative extinction due to the contrail does not only depend on the optical properties of the contrail itself, but also on the amount of radiation, irradiating the contrail. The powers of solar intensities and terrestrial irradiances depend on wavelength, longitude, latitude, altitude, the presence of clouds, time of the day and time of the year. Most of these input variables are provided by the flight performance model (compare Fig. 1). For the calculation of this "Atmospheric Radiative Transfer", the radiative transfer software package libRadtran [46] is used. Due to different properties of solar and terrestrial extinction in the atmosphere, different radiative transfer solvers are used.

*1) Terrestrial Radiative Transfer:* In the terrestrial wavelength spectrum (3  $\leq \lambda \leq 100 \mu m$ ), absorption by atmospheric molecules strongly depends on wavelength and varies between neighboring wavelengths. These narrow absorption bands require a high spectral resolution in the radiative transfer calculation and therefore a high computational effort. However, a weak angular dependence (described by zenith angle θ and azimuthal angle φ) of radiation due to a missing part of direct irradiance is expected. The Two Stream Approximation (TSA) takes advantage of the weak angular dependence and reduces the computational effort [47]. Here, all shares of radiation coming from one hemisphere are azimuthally averaged over the half space and are treated as a single irradiance  $F$  [W m<sup>-2</sup>] without specific information about the angular direction (compare Fig. 6, left). Due to this average, two irradiances at any altitude remain: terrestrial irradiances coming from the lower hemisphere

$$
F_{\text{up}}(\lambda, t, \theta = \frac{1}{2}\pi \dots \frac{3}{2}\pi) \tag{3}
$$

and terrestrial irradiances coming from the upper hemisphere

$$
F_{\text{down}}(\lambda, t, \theta = \frac{3}{2}\pi...^{\frac{1}{2}}\pi)
$$
 (4)

(compare Fig. 6, left) [11].



Figure 6. Azimuthally averaged terrestrial irradiances  $F_{\text{up}}$  and  $F_{\text{down}}$  (left) and angular dependent diffuse solar intensities  $I_{\text{diff}}(\theta, \phi)$ . The direct beam  $I<sub>dir</sub>$  coming from the sun is added to the corresponding solid angle of the position of the sun.

*2) Solar Radiative Transfer:* In the solar wavelength spectrum  $(0 \le \lambda \le 4 \mu m$ , compare Fig. 7) a TSA is out of question, because of a large influence of the direct beam (in the direction of the position of the sun) on the radiation field. This influence causes a strong angular radiative dependence which cannot be described by a TSA. The radiative transfer solver DISORT (DIScrete Ordinate Radiative Transfer solver) is used for the angular dependent calculation of direct (5) solar intensities (5)  $\text{[mW sr}^{-1} \text{m}^{-2} \text{nm}^{-1}$  calculated.

$$
I_{\text{dir}}(\lambda, t, \text{lon}, \text{lat}, \Omega) \tag{5}
$$

Here, the direction of the direct beam,  $\Omega$  [sr], is described by the solar zenith angle and the solar azimuthal angle. Diffuse solar intensities (6)  $\text{[mW sr}^{-1}\text{m}^{-2}\text{nm}^{-1}\text{]}$ 

$$
I_{\text{diff}}(\lambda, t, \text{lon}, \text{lat}, \theta, \phi) \tag{6}
$$

depending on longitude, latitude, altitude, time of the day and time of the year [48] are also provided by DISORT. DISORT is the most used, recommended and most updated solver for angular depending radiative transfer in the shortwave spectrum [11, 49]. Diffuse solar intensities  $I_{\text{diff}}$  are calculated with an angular discretization of  $d\theta = d\phi = 2^{\circ}$  and the direct beam  $I_{\text{dir}}$  is added to the solid angle  $d\omega$  [sr]

$$
d\omega = \sin \theta \, d\theta \, d\phi,\tag{7}
$$

where  $\theta$  and  $\phi$ , are the solar zenith and azimuthal angle, respectively (compare Fig. 6, right) [11]. Fig. 7 clearly indicates the different contributions of wavelength-specific irradiances on the radiation budget of the Earth-Atmosphere System. Hence, a consideration of solar irradiances around the maximum at  $\lambda = 0.55$  (i.e.  $0.2 < \lambda < 1$   $\mu$ m) would be sufficient. Terrestrial irradiances should be considered between  $5 < \lambda < 22 \mu m$ . Additionally, Fig. 7 proves the significant impact of atmospheric absorption by comparing the modeled irradiances with the theoretical values, calculated with Planck's function assuming mean temperatures of 5750 K of the sun and 288 K of the Earth's surface.

## *F. Contrail Radiative Forcing*

The radiative quantities of the "Atmospheric Radiative Transfer" model are multiplied with the extinguished number ratios, provided by the "Contrail Optical Properties" model to estimate the wavelength specific and direction specific extinction of radiation due to the contrail. Because the contrail radiative forcing is defined as imbalance of the radiation budget at a horizontal layer, we distinguish between sources and drains of radiation in the upper and in the lower hemisphere of the contrail (compare Fig. 8).

To calculate the radiative forcing per unit length of the contrail, the power  $P_i(\lambda, t, \text{lon}, \text{lat}, d\omega)$  [W m<sup>-1</sup> nm<sup>-1</sup>] of the extinguished photons once irradiated on a unit length contrail have to be considered and balanced. Therefore, the solar intensities (6) and the terrestrial irradiances (3) and (4) coming from a particular solid angle  $d\omega$  [sr], (calculated



Figure 7. Modeled and approximated (as Planck's function) solar and terrestrial irradiances over Berlin, Germany in 10000 m altitude. The extinction of radiation by the atmosphere (without contrail) is remarkable for solar wavelengths  $\lt 0.2 \mu m$  and for the whole terrestrial wavelength spectrum.

by the "Atmospheric Radiative Transfer" model) has to be weighted by the weighted number ratios of extinguished photons  $S_i$  [m] (2) (from the "Contrail Optical Properties" model) and the corresponding solid angle  $d\omega_i$ . Therefore, only the backscattered and absorbed powers are of interest and the extinguished powers can be summarized to the following components (compare Fig. 8):

- $P_{\uparrow \downarrow a}$ : number of absorbed photons coming from below
- $P_{\uparrow b}$ : number of photons coming from below, scattered into the lower hemisphere
- $P_{\perp b}$ : number of photons coming from above, scattered into the upper hemisphere

For radiation coming from a particulate solid angle  $d\omega_i$  (7) and getting scattered into the same hemisphere (i.e. backscattered) the extinguished power  $P_b(\lambda, t, \text{lon}, \text{lat}, d\omega_i)$  [W m<sup>-1</sup>nm<sup>-1</sup>] is

$$
P_{\downarrow b}(\lambda, t, d\omega_i) = I(\lambda, t, d\omega_i) \cdot S_b(\lambda, t, d\omega_i) \cdot d\omega_i
$$
  
\n
$$
P_{\uparrow b}(\lambda, t, d\omega_i) = I(\lambda, t, d\omega_i) \cdot S_b(\lambda, t, d\omega_i) \cdot d\omega_i
$$
  
\n
$$
P_{\downarrow\uparrow a}(\lambda, t, d\omega_i) = I(\lambda, t, d\omega_i) \cdot S_a(\lambda, t, d\omega_i) \cdot d\omega_i,
$$

where  $S_b(\lambda, t, d\omega_i)$  denotes the weighted number ratio of backscattered photons coming from a particular solid angle  $d\omega_i$  and getting scattered into the same hemisphere and  $I(\lambda, t, d\omega_i)$  is the wavelength specific solar intensity from the particular solid angle  $d\omega_i$ . In the same way, absorbed solar powers  $P_{\text{ln}}(\lambda, t, \text{lon}, \text{lat}, d\omega_i)$  are calculated for each wavelength  $\lambda$  and each solid angle  $d\omega_i$ .

Terrestrial irradiances  $F_{\text{up}}$  and  $F_{\text{down}}$  [W  $\text{m}^{-2}\text{nm}^{-1}$ ], calculated with the Two Stream Approximation are hemispherically averaged irradiances. To estimate the extinguished power in the terrestrial wavelength spectrum,  $F_{\rm up}$  and  $F_{\rm down}$  have to weighted by the number ratio of extinguished photons coming from the upper hemisphere  $S_{\downarrow b}$  and the lower hemisphere  $S_{\uparrow b}$  both getting scattered into the same hemisphere as they are coming from and the absorbed photons coming from all directions  $S_{\downarrow\uparrow a}$ . For example:

$$
P_{\downarrow b}(\lambda, t) = F_{\text{down}} S_{\downarrow b}(\lambda, t) \tag{8}
$$

$$
P_{\uparrow b}(\lambda, t) = F_{\rm up} S_{\uparrow b}(\lambda, t) \tag{9}
$$

$$
P_{\downarrow\uparrow a}(\lambda, t) = F_{\text{down}}(\lambda, t) S_{\downarrow a}(\lambda, t)
$$
 (10)

$$
+ F_{\rm up}(\lambda, t) \, S_{\uparrow a}(\lambda, t). \tag{11}
$$

The backscattered powers  $P_b(\lambda, t, \text{lon}, \text{lat}, d\omega)$  are categorized into two classes: first, backscattered powers from the upper hemisphere  $P_{\downarrow b}(\theta = \frac{3}{2}\pi...\frac{1}{2}\pi)$  (resulting in a cooling effect) and second from the lower hemisphere  $P_{\uparrow b}(\theta = \frac{1}{2}\pi...\frac{3}{2}\pi)$ (with a heating effect, compare Fig. 8). Absorbed powers  $P_{\text{ln}a}(\lambda, t, \text{lon}, \text{lat}, d\omega)$  coming from both hemispheres always contribute to a contrail heating. Those components contribute to the radiative forcing  $RF_{\text{Contrail}}$ 

$$
RF_{\text{Contrail,m}} = P_{\downarrow\uparrow a} + P_{\uparrow b} - P_{\downarrow b}.
$$
 (12)



Figure 8. Components of backscattered and absorbed powers contributing to the contrail radiative forcing (compare (12)). Blue parts denote cooling effects, red parts indicate warming effects.

After integrating over all significant wavelengths, time steps of the contrail life cycle and over the contrail length, each with different optical properties  $S_i(\lambda, t)$ , the total radiative forcing  $RF_{\text{Contrail,m}}$  [W m<sup>-1</sup>] (12) per meter of an individual contrail can be estimated. By multiplying (12) with the length  $L_{\text{Contrail}}$  of the contrail with similar optical properties offers the extinguished power [W] due the contrail. Usually, the radiative forcing is related to the Earth's surface, which is  $A_{\text{Earth}} = 5.1 \cdot 10^{14} \text{ m}^2$ . Hence, the estimated extinguished power is divided by  $A_{\text{Earth}}$ .

$$
RF_{\text{Contrail}} = \frac{RF_{\text{Contrail,m}} L_{\text{Contrail}}}{A_{\text{Earth}}}.\tag{13}
$$

#### *G. Weighting of Contrail Costs in Trajectory Optimization*

The integration of  $RF_{\text{Contrail}}$  over the whole life cycle, over all significant wavelengths and the division by  $A_{\text{Earth}}$  offers a comparison with the impact of a reference emission (e.g.  $CO<sub>2</sub>$ ) over the whole flight.

Usually, a certain time horizon H (e.g.  $H = 100$  years) is considered in those interpretations. In this case, the ratio of  $RF_{\text{Contrail}}$  over  $RF_{\text{CO}_2}$  (known as the global warming potential GWP)

$$
GWP = \frac{\int_{0}^{H} RF_{\text{Contrail}}(t)dt}{\int_{0}^{H} RF_{\text{CO}_2}(t)dt}
$$
(14)

is used to calculate  $CO<sub>2</sub>$  equivalent emissions, which are transferred into monetary values by applying political instruments, such as the emission trading scheme ETS. For instance, for  $H = 100$  years, the radiative forcing of the total amount of  $CO<sub>2</sub>$  in the atmosphere is [50]

$$
\int_{0}^{H=100} R F_{\text{CO}_2}(t) dt = \alpha_{\text{CO}_2} \ln \left( \frac{\text{C}}{\text{C}_0} \right) = 1.94 \text{ W m}^{-2} \quad (15)
$$

where  $\alpha_{\text{CO}_2}$  = 5.35 denotes a constant [50], C = 399.39 ppm is the actual concentration of  $CO<sub>2</sub>$  [51] and  $C_0 = 278$  ppm is the pre-industrial concentration of  $CO_2$ in 1959 [51].

Equation (15) considers the total sum of  $CO<sub>2</sub>$  emissions in 100 years and is not restricted to the aviation transport sector. The radiative impact of  $CO<sub>2</sub>$  strongly depends on the altitude of emission. In high altitudes (above the tropopause with low vertical exchange)  $CO<sub>2</sub>$  exists for 400 years. In low altitudes (below the tropopause) precipitation induces washing-off effects of  $CO<sub>2</sub>$ . Those large differences in the residence time, complicate the use of (15) for a comparison of aviation induced radiative forcing due to contrails and  $CO<sub>2</sub>$ emissions.

However, from the literature review in the introduction of this paper we know the amount of  $CO<sub>2</sub>$  emissions, caused by aviation in 2005

$$
m_{\text{CO}_2,2005} = 733 \ 10^6 \ \text{ta}^{-1} \tag{16}
$$

[7] and the radiative forcing of those emissions

$$
RF_{\text{CO}_2,2005} = 0.028 \text{ W m}^{-2}.
$$
 (17)

Hence, the impact of each tonne  $CO<sub>2</sub>$  emitted by aviation in 2005 was

$$
RF_{\rm CO_2} = 3.8 \ 10^{-11} \ \rm W \, \rm m^{-2} \rm t_{\rm CO_2}^{-1}.
$$
 (18)

Therewith, the radiative forcing of an individual contrail (13) can be transformed into tonnes of  $CO<sub>2</sub>$  equivalent emissions, which can be used as external costs in a multi-criteria trajectory optimization.

$$
m_{\text{CO}_2\text{eq}} = \frac{RF_{\text{Contrail}}}{3.8 \ 10^{-11} \ \text{W} \ \text{m}^{-2} \text{t}^{-1}_{\text{CO}_2}}. \tag{19}
$$

By weighting the price per tonne of  $CO<sub>2</sub>$  equivalent emission, contrails are considered in the multi-criteria trajectory optimization TOMATO. Following global approximations, as elaborated in the introduction, it is expected, that the radiative impact of a single flight's contrail exceeds those effects of  $CO<sub>2</sub>$ .

## III. RESULTS

The most important effects of  $RF_{\text{Contrail}}$  are determined by the angle  $\alpha$  between the length axis of the contrail and the incoming photons (compare Fig. 4), because it determines the travel distance of photons through the contrail. Furthermore,  $RF_{\text{Contrail}}$  depends on the order of magnitude of incoming radiation, depending on wavelength (solar  $>$  terrestrial), time of the day and year (summer, noon  $>$  winter, morning and evening) and on latitude (equator  $>$  pole). Both effects are shown in Fig. 9, because  $\alpha$  and solar intensity change with daytime. Although the amount of backscattered power  $P_{\downarrow b}$ dominates over the whole day, the sum of downward backscattered power  $P_{\uparrow b}$  and absorbed power  $P_{\downarrow \uparrow a}$  nearly compensates  $P_{\perp b}$  between 7 a.m. and 5 p.m. Only during sunrise and sunset with horizontal photon transport (around 6 a.m. and 6 p.m.),  $RF_{\text{Control}}$  is significantly negative. For astrologic reasons, the "sunrise-sunset effect" increases with increasing latitude, but the amplitude in Fig. 9 decreases with increasing latitude.



Figure 9. Left: Backscattered power for diurnal variations of upward  $(P_{\uparrow b})$ and downward ( $P_{\perp b}$ ) solar radiation ( $\lambda = 0.55 \mu$ m) and absorbed power of downward and upward radiation  $(P_{\downarrow} \uparrow a)$  for a contrail in June over Berlin, Germany, which constitutes the North-South axis. Right: resulting  $RF_{\text{Contrail}}$ (12) from the components shown left.

Although in Fig. 9 the extinguished of only a single wavelength is shown, the results my be transferred to the narrow wavelength band around  $0.2 < \lambda < 2$   $\mu$ m. In Fig. 10, simulations of the whole solar spectrum with significant contribution to the energy budget with wavelengths between 0.55  $\mu$ m  $\leq$  $\lambda$  < 4.5  $\mu$ m (left) and of the most important terrestrial spectrum wavelengths between 4.5  $\mu$ m  $\leq \lambda \leq 21.5 \mu$ m (right) are shown. Fig. 10, left, indicates a decreasing solar effect with increasing wavelength due to a decreasing solar intensity and due to an increasing share of absorbed power. However, with increasing wavelength, positive values of  $RF_{\text{contrail}}$  become more dominant. Note, in Fig. 9 and 10 only the radiative properties of a single contrail with specific optical properties are calculated. The absorption efficiency  $Q_a$ , known as the possibility of an absorbing event within the contrail strongly varies with ice particle size and shape, which is why the fluctuation in Fig. 10 (left) is very contrail-specific.

The strong impact of ice particle size on  $Q_a$  causes an increasing terrestrial radiative forcing with increasing contrail lifetime (compare Fig. 11 and Fig. 2 and 3 for increasing ice particle radius with lifetime). Additionally, an increasing contrail width with life time causes a more distinct radiative forcing of older contrails. Note, the solar radiative forcing (Fig. 11,



Figure 10. Wavelength dependent contrail radiative forcing of a single contrail over Berlin, Germany. In Mid-latitudes, the impact of solar extinction is more significant than the terrestrial contribution to  $RF_{\rm Contrail}$ . With increasing wavelength  $RF_{\text{Contrail}}$  converges to a balanced positive value.

left) is shown, only for a single wavelength  $\lambda = 0.55 \mu m$ , whereas the terrestrial  $RF_{\text{Contrail}}$  is integrated over the whole significant terrestrial spectrum 4.5  $\mu$ m  $\leq \lambda \leq 21.5 \mu$ m. The large computational effort requires the reduction of the number of calculations, especially in the solar wavelength spectrum.



Figure 11. Solar  $RF_{\text{Contrail}}$  at  $\lambda = 0.55 \ \mu \text{m}$  depending on the life cycle of single contrail (left). Right: increasing terrestrial  $RF_{\rm Contrail}$  with contrail life time, integrated over the whole significant terrestrial spectrum 4.5  $\mu$ m  $\leq$  $\lambda \leq 21.5 \mu m$ .

Finally, the impact of the consideration of contrails in trajectory optimization is exemplified in Fig. 12. Here, blue squares denote ice-supersaturated regions wherein contrail formation is very likely. For this example flight from Los Angeles (LAX) to Boston (BOS) different routes are calculated and compared with each other. The originally filed route (black, 4580 km ground distance ) induced a contrail with a length of  $L_{\text{Contrail}} = 1705$  km ground distance. The mean optical properties of this contrail have been estimated to  $RF_{\text{Contrail}} =$ 0.305 W m<sup>-1</sup> which is 3.1 10<sup>-6</sup>W m<sup>-2</sup>. Following (19), this contrail must be weighted with  $81578 \text{ t } CO_2$  equivalent emissions. Considering this weighting in a multi-criteria trajectory optimization, the contrail would have been completely avoided (red route in Fig. 12 with 4730 km ground distance). A more harmonized trajectory with a reduced contrail length of  $L_{\text{Contrail}} = 1374$  km ground distance constitutes the green route in Fig. 12 with a total ground distance of 4246 km and a contrail weighting of  $66087$  t  $CO<sub>2</sub>$  equivalent emissions.

## IV. CONCLUSIONS

In this study, a method is described to consider the environmental impact of individual condensation trails in a multicriteria trajectory optimization. A tendency towards lower



Figure 12. Lateral routes of optimized trajectories between Los Angeles and Boston with different contrail weightings in an ice-supersaturated atmosphere (blue squares). Significant differences between the filed route (black), a weighting of 32 tonnes  $CO<sub>2</sub>$  (green) and a weighting of 40 tonnes  $CO<sub>2</sub>$ (red) per hour contrail formation leads to a complete contrail avoidance.

aircraft speeds and higher fuel flows has been identified to decrease the probability of contrail formation. The atmospheric parameters wind shear, vertical wind speed and turbulence mainly influence the initial contrail dimensions (weak turbulence: small contrail) and the contrail life time (strong turbulence: short life time). To consider multiple scattering events a Monte Carlo Simulation is neccessary, where the position and the kind of extinction event are determined probabilistically. The radiative extinction due to the contrail is calculated separately from the atmospheric radiative extinction. The advantage of this approach is, that radiative properties of optically similar contrails (calculated in the "Contrail Optical Properties" model) can be easily combined with different atmospheric conditions (estimated in the "Atmospheric Radiative Transfer" model), which are a function of the position, time of the day and year. In Mid-Latitudes, even during daytime the warming effect of the contrail dominates, mainly because of an increasing absorbed power with increasing contrail life time in the terrestrial wavelength spectrum. A significant impact of the track angle (i.e. the contrail length axis) on the radiative properties has been elaborated because it determines the travel distance of photons during sunrise and sunset. However, suggestions of preferring North-South oriented routes to East-West routes is less helpful in daily operations.

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